Buoyancy Effects on Laminar Mixed Convection in Vertical Channel With Dissipation

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Abstract: Deals with the buoyancy effects on laminar mixed convection in vertical channel is considered. Solutions of the governing parabolic equations are obtained by the use of an implicit finite difference technique coupled with a marching procedure. The velocity, the temperature and the pressure profiles are presented graphically for different values of governing parameters like Eckert Number Ek, buoyancy parameter Gr/Re and Prandtl number Pr and their behaviour discussed.

Keywords: Mixed convection, Vertical channel, Dissipation

I. Introduction

Several authors have presented theoretical or experimental investigations most of which have been reviewed, for instance, in Aung [1]. The literature of the last decades includes many papers presenting theoretical investigations of buoyancy-induced flows in vertical or inclined ducts. Some of these papers (Barletta and Zanchini [7]; Chamkha [9]) describe analytical solutions with reference to fully developed parallel flows. Infact, parallel flow represents the condition that allows a drastic simplification of the governing balance equations, thus giving a chance to find exact solutions at least for the simpler geometries: parallel plane channel, circular or annular duct, rectangular duct. However, if one deals with inclined ducts, the condition of parallel flow can be considered as an exception rather than a rule. This conclusion is a direct consequence of the observation that, in an inclined duct, the buoyancy force vector has a non-vanishing projection on the plane of the duct cross-section. In the fully developed region, the transversal components of the buoyancy force are normally responsible for the onset of a secondary flow that makes velocity a non-parallel helicoidal vector field. It must be pointed out that, in this reasoning, geometry matters. In fact, for a parallel plane channel, parallel flows are still possible when the channel is inclined, for the more commonly employed thermal boundary conditions (Barleta and Zanchini [7]). For different geometries of the duct cross-section, as circular, annular or rectangular, this is not true.

With reference to mixed convection in vertical or inclined plane channels, the fully developed regime has been studied analytically by Aung and Worku [3], Barletta and Zanchini [5, 6]; the flow stability has been investigated numerically by Chen and Chung [10, 11]; the reversed flow has been studied experimentally by Gau et al. [13]. Buoyancy effects on Laminar Mixed convection in vertical channels is studied by Reddy [14]. Aung and Worku [2] were discussed the results of a study laminar mixed convection in a parallel plate vertical channel in which the two walls are maintained at uniform but not necessarily equal temperatures. Barletta [4] have been studied laminar mixed convection in a vertical rectangular duct. The analysis refers to thermal boundary conditions such that atleast one of the four duct walls is kept iso-thermal. The evaluation of the velocity field and of the temperature field is performed analytically. Fully developed mixed convection and flow reversal in a vertical rectangular duct with uniform wall heat flux was studied by Barletta [7]. Yih et al. [15] studied numerically heat transfer enhancement in vertical parallel channel with asymmetric walls. Evans and Greif [12] were studied the buoyant in stabilities in downward flow in a symmetrically heated vertical channel.

II. Nomenclature

- b Distance between the parallel plates
- C_p Specific heat of the fluid
- Ek Eckert Number
- g Gravitational body force per unit mass (acceleration)
- Gr Grashoff number, $(g\beta (T_1 T_0) b^3 / v^2)$
- k Thermal conductivity of fluid
- p Local pressure at any cross section of the vertical channel
- P Dimensionless pressure inside the channel at any cross section, $(p p_0) / \rho u_0^2$

- Pr Prandtl Number, $(\mu C_p / k)$
- Re Reynolds number, $((b u_0)/\upsilon)$
- T Dimensional temperature at any point in the channel
- T₀ Ambient of fluid inlet temperature
- T_w Isothermal temperatures of circular heated wall
- T_1,T_2 Isothermal temperatures of plate 1 and plate 2 of parallel plates
- u Axial velocity component
- \overline{u} Average axial velocity
- u₀ Uniform entrance axial velocity
- $U \qquad \ \ \, Dimensionless \ axial \ velocity \ at \ any \ point, \ u \ / \ u_0$
- v Transverse velocity component
- V Dimensionless transverse velocity, v / u_0
- x Axial coordinate (measured from the channel entrance)
- X Dimensionless axial coordinate in Cartesian, (x / (b Re))
- y Transverse coordinate of the vertical channel between parallel plates
- Y Dimensionless transverse coordinate, y / b
- θ Dimensionless temperature
- ρ Density of the fluid
- ρ_0 Density of the fluid at the channel entrance
- μ Dynamic viscosity of the fluid
- v Kinematic viscosity of the fluid, μ/ρ
- β Volumetric coefficient of thermal expansion

III. Formulation Of The Problem

We consider the study flow of an incompressible viscous fluid on developed laminar mixed convection in vertical channels. Figure 1 depict two-dimensional channel between two vertical parallel plates. The distance between the plates is 'b' i.e., the channel width. The Cartesian coordinate system is chosen such that the x-axis is in the vertical direction that is parallel to the flow direction and the gravitational force 'g' always acting downwards independent of flow direction. The y-axis is orthogonal to the channel walls, and the origin of the axes is such that the positions of the channel walls are y=0 and y=b. One wall is isothermal and other wall is adiabatic condition.

The governing equations for the study viscous flow with the following assumptions are made:

- (i) The flow is steady, viscous, incompressible, and developed.
- (ii) The flow is assumed to be two-dimensional steady, and the fluid properties are constant except for the variation of density in the buoyancy term of the momentum equation.

After applying the above assumption the boundary layer equations appropriate for this problem are Continuity Equation

$$\frac{\partial U}{\partial X} + \frac{\partial V}{\partial Y} = 0$$

Momentum Equation

$$U\frac{\partial U}{\partial X} + V\frac{\partial U}{\partial Y} = -\frac{dP}{dX} + \frac{Gr}{Re}\theta + \frac{\partial^2 U}{\partial Y^2}$$

Energy Equation

Energy Equation

$$U\frac{\partial\theta}{\partial X} + V\frac{\partial\theta}{\partial Y} = \frac{1}{\Pr}\frac{\partial^2\theta}{\partial Y^2} + Ek\left(\frac{\partial U}{\partial Y}\right)^2$$



Figure 1: Schematic view of the system and coordinate axes corresponding to Up-flow

(1)

(2)

(3)

The form of continuity equation can be written in integral form as

Gr = g β (T₁ - T₀) b³/ ν ², θ = (T - T₀)/(T₁ - T₀)

$$\int_{0}^{1} U \, dY = 1$$
(4)
The boundary conditions are
Entrance conditions
At X = 0, 0 < Y < 1 : U = 1, V = 0, $\theta = 0, P = 0$
(5a)
No slip conditions
At X > 0, Y = 0 : U = 0, V = 0
At X > 0, Y = 1 : U = 0, V = 0
Thermal boundary conditions
At X > 0, Y = 1 : $\theta = 1$
At X > 0, Y = 1 : $\left(\frac{\partial \theta}{\partial Y}\right)_{Y=1} = 0$
(5b)
In the above dimensionless parameters have been-defined as:
U = u / u_0, V = vb / v, X = x / (b Re), Y = y / b
P = (p - p_0) / \rho u_0^2, Pr = \mu C_p / k, Re = (b u_0) / v
(6)

The systems of non-linear equations (1) to (3) are solved by a numerical method based on finite difference approximations. An implicit difference technique is employed whereby the differential equations are transformed into a set of simultaneous linear algebraic equations.

IV. Numerical Solution

The solution of the governing equations for developing flow is discussed in this section. Considering the finite difference grid net work of figure 2, equations (2) and (3) are replaced by the following difference equations which were also used in [8].



Numerical representation of the Integral Continuity Equation

The integral continuity equation can be represented by the employing a trapezoidal rule of numerical integration and is as follows:

$$\left[\sum_{j=1}^{n} U(i+1, j) + 0.5(U(i+1, 0) + U(i+1, n+1))\right] \Delta Y = 1$$

However, from the no slip boundary conditions U(i+1, 0) = U(i+1, n+1) = 0

Therefore, the integral equation reduces to:

$$\left[\sum_{j=1}^{n} U\left(i+1, j\right)\right] \Delta Y = 1 \tag{9}$$

A set of finite-difference equations written about each mesh point in a column for the equation (7) as shown: $RU(i+11) + \alpha U(i+12) + \beta P(i+11) + \frac{Gr}{2} Q(i+11) = 4$

$$\rho_{1}U(i+1,1) + \gamma_{1}U(i+1,2) + \zeta P(i+1) + \frac{Gr}{Re}\theta(i+1,1) = \phi_{1}$$

$$\alpha_{2}U(i+1,1) + \beta_{2}U(i+1,2) + \gamma_{2}U(i+1,3) + \zeta P(i+1) + \frac{Gr}{Re}\theta(i+1,2) = \phi_{2}$$

$$\alpha_n U(i+1,n-1) + \beta_n U(i+1,n) + \xi P(i+1) + \frac{Gr}{\text{Re}} \theta(i+1,n) = \phi_n$$

where

$$\alpha_{k} = \frac{1}{\left(\Delta Y\right)^{2}} + \frac{V(i, j)}{2\Delta Y}, \quad \beta_{k} = -\left[\frac{2}{\left(\Delta Y\right)^{2}} + \frac{U(i, j)}{\Delta X}\right]$$
$$\gamma_{k} = \frac{1}{\left(\Delta Y\right)^{2}} - \frac{V(i, j)}{2\Delta Y}, \quad \xi = \frac{-1}{\Delta X}, \qquad \phi_{k} = -\left[\frac{P(i) + U^{2}(i, j)}{\Delta X}\right]$$

for k = 1, 2... n

A set of finite-difference equations written about each mesh point in a column for the equation (8) as shown: $\overline{B}_{i}\theta(i+1,1) + \overline{\nu}_{i}\theta(i+1,2) + \dots + \overline{\mu}_{i}\theta(i+1,2) + \dots + \overline{\mu}_$

$$\overline{\alpha}_{2}\theta(i+1,1) + \overline{\beta}_{2}\theta(i+1,2) + \overline{\gamma}_{2}\theta(i+1,3) + \dots = \overline{\phi}_{1} \quad \alpha_{1}$$

$$\overline{\alpha}_{2}\theta(i+1,1) + \overline{\beta}_{2}\theta(i+1,2) + \overline{\gamma}_{2}\theta(i+1,3) + \dots = \overline{\phi}_{2}$$

$$\overline{\alpha}_{n}\theta(i+1,n-1) + \overline{\beta}_{n}\theta(i+1,n) + \overline{\gamma}_{n}\theta(i+1,n+1) + \dots = \overline{\phi}_{n}$$

$$(\overline{\alpha}_{n+1} + \overline{\gamma}_{n+1})\theta(i+1,n) + \overline{\beta}_{n+1}\theta(i+1,n+1) = \overline{\phi}_{n+1}$$
where
$$1 = V(i,j)$$

$$\overline{\alpha}_{k} = \frac{1}{\Pr\left(\Delta Y\right)^{2}} + \frac{V(i, j)}{2\Delta Y}, \qquad \overline{\beta}_{k} = -\left\lfloor \frac{2}{\Pr\left(\Delta Y\right)^{2}} + \frac{U(i, j)}{\Delta X} \right\rfloor, \qquad \overline{\gamma}_{k} = \frac{1}{\Pr\left(\Delta Y\right)^{2}} - \frac{V(i, j)}{2\Delta Y},$$
$$\overline{\phi}_{k} = -\frac{U(i, j)\theta(i, j)}{\Delta X} - Ek\left(\frac{U(i+1, j+1) - U(i+1, j-1)}{2\Delta Y}\right)^{2}$$
for k = 1.2 n+1

Equation (12) can be written as

$$V(i+1,j) = V(i+1,j-1) - \frac{\Delta Y}{2\Delta x} (U(i+1,j) + U(i+1,j-1) - U(i,j) - U(i,j-1)), j = 1,2,\dots,n \quad (10)$$

The numerical solution of the equations is obtained by first selecting the parameter that are involved such as Gr/Re, M and Pr. Then by means of a marching procedure the variables U, V, θ and P for each row beginning at row (i+1) = 2 are obtained using the values at the previous row 'i'. Thus, by applying equations (7), (8) and (9) to the points 1, 2, ..., n on row i, 2n+2 algebraic equations with the 2n+2 unknowns U(i+1,1), U(i+1,2), ..., U(i+1,n), P(i+1), $\theta(i+1,1), \theta(i+1,2), \ldots, \theta(i+1, n), \theta(i+1, n+1)$ are obtained. This system of equations is then solved by Gauss – Jordan elimination method. Equations (10) are then used to calculate V(i+1,1), V(i+1,2), ..., V(i+1,n).

V. Results And Discussion

The numerical solution of the equation is obtained first selecting the parameters that are involved such as Gr/Re, Ek and Pr. For fixed Pr=0.7 and different values of Gr/Re and Ek, the velocity profiles are shown in figures 3 to 5, the temperature profiles are shown in figures 6 to 8 and pressure profiles are shown in figures 9 to 11.

Figures 3 to 5 show the velocity profiles for fixed X as can be seen, vary closed to the channel inlet, the heating effects are not yet felt and further downstream, the heating causes the fluid accelerate near the hot wall and decelerate near the unheated wall (adiabatic wall) resulting in distortion of velocity profiles to satisfy the continuity principle. With further increase in heating rates, this distortion becomes more severe and results into flow reversal near the adiabatic wall. However the velocity profile recovers and attains it asymptotic fully developed parabolic profile. A skewness in the velocity profiles also appear as the fluid moves towards the hot wall (Y=1). The velocity profiles are not effect for increasing Eckert number Ek and for fixed Prandtl number Pr and buoyancy parameter Gr/Re.

Figures 6 to 8 shows the development of temperature profile for fixed buoyancy parameter Gr/Re and Eckert number Ek. These figures show how the temperature profiles are developing approaching there invariant analytical fully developed profiles. For the asymmetric heating conditions the temperature profiles will developed from the entrance till approach it's a linear profiles of the fully developed region where the fluid flows in laminated layers and all the heat transferred from the heated wall goes to the cold wall through the fluid laminated layers. It is seen that the temperature profiles are effect for increasing Eckert number Ek and for fixed Gr/Re and Prandtl number Pr at r_T for different fixed X values.

The development of pressure profiles are shown in figures 9 to 11 for buoyancy aided flow. These figures shows how the two hydrodynamic parameters are developing downstream of the channel at different heating rates represented buoyancy parameter Gr/Re. These negative values of the pressure will continue increasing due to the negative pressure gradient for the cases of pure forced convection (Gr/Re = 0). The pressure profiles are not effect for increasing Eckert number Ek and for fixed Prandtl number Pr and buoyancy parameter Gr/Re.









Figure 6(b): Temperature profile for fixed X = 0.04and Ek = 1



Figure 6(d): Temperature profile for fixed X = 0.04and Ek = 10



Figure 7(b): Temperature profile for fixed Gr/Re =50 and Ek =1 $\,$



Figure 6(c): Temperature profile for fixed X = 0.04 and Ek = 5



Figure 7(a): Temperature profile for fixed Gr/Re =50 and Ek = 0



Figure 7(c): Temperature profile for fixed Gr/Re =50 and Ek = 5 $\,$



Figure 7(d): Temperature profile for fixed Gr/Re =50 and Ek = 10



Figure 8(b): Temperature profile for fixed Gr/Re



Figure 8(d): Temperature profile for fixed Gr/Re =200 and Ek =10



Figure 8(a): Temperature profile for fixed Gr/Re =200 and Ek = 0







Figure 9: Pressure profile



Figure 10: Pressure profile

Figure 11: Pressure profile

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